CLAIMS

We claim:

A fluidic device for an aircraft comprising:

 a passage block defining an internal space;
 an internal body, an outer surface of the internal body being liquid tightly fitted to the internal space;

an actuator having a piston and a cylinder chamber; and a fluid-controlled valve; wherein the passage block defines:

first fluid passages communicating the internal space with the cylinder chamber of the actuator,

second fluid passages communicating the internal space with the fluid-controlled valve, and the internal body defines grooves by which the first and second fluid passages are communicated, the grooves extending along the outer surface of the internal body.

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